

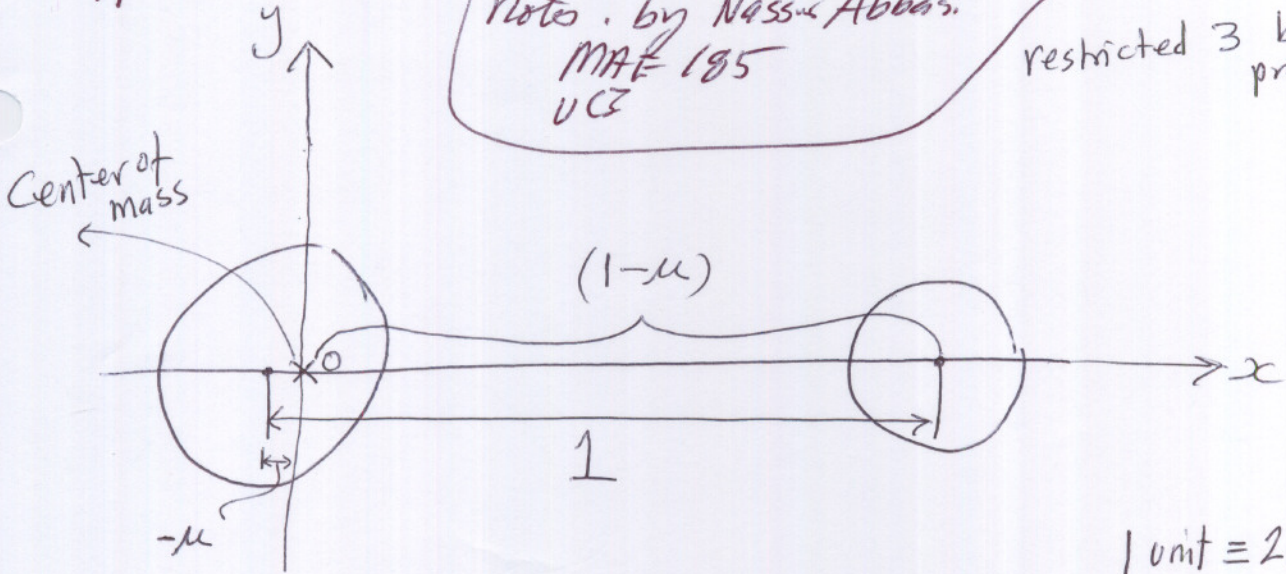
HW#2

Friday April 28 discussion

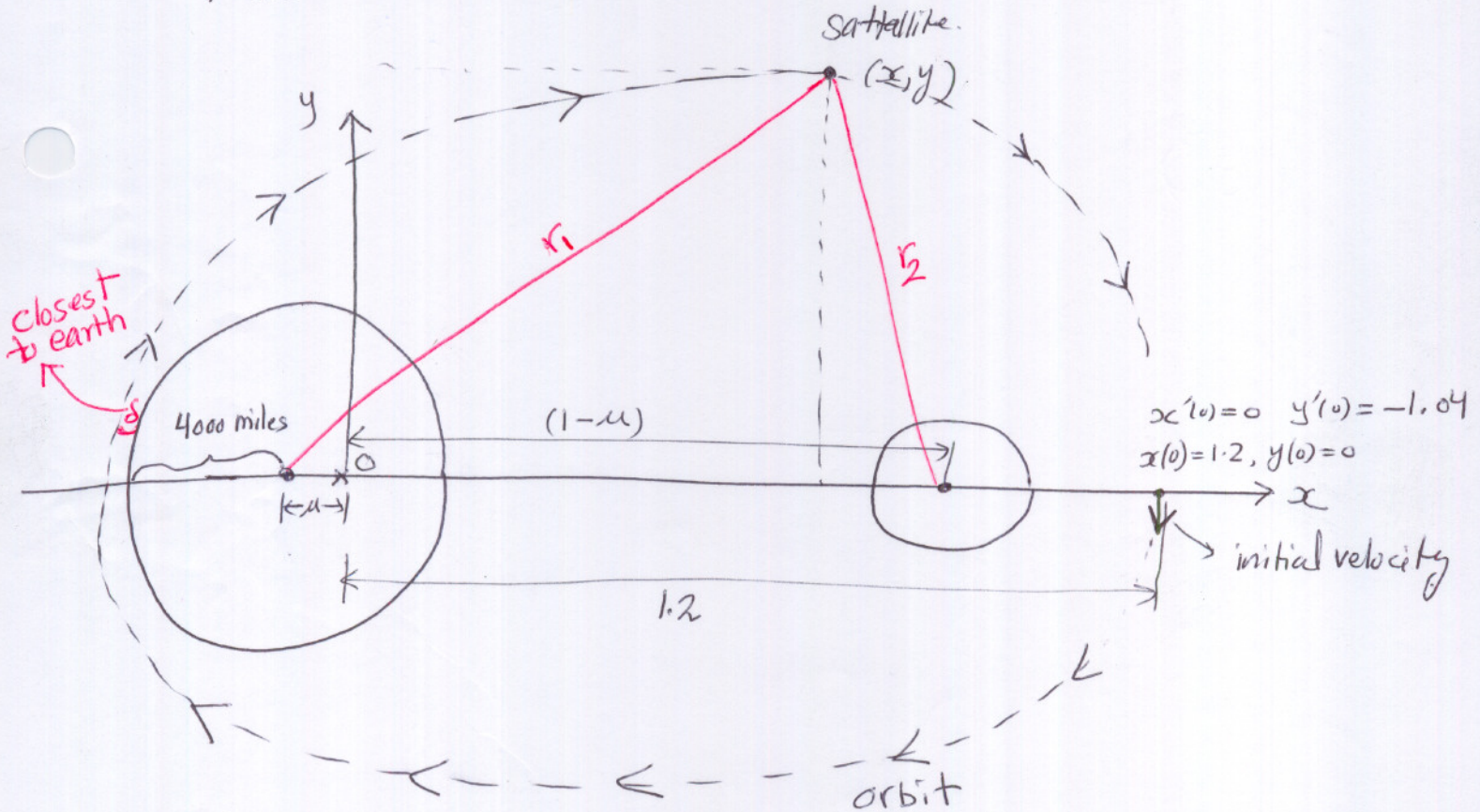
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restricted 3 body problem.

①



$$\mu = \frac{\text{Mass Moon}}{\text{Mass Earth}}$$



Step 1 Convert 2 Second order ODE's to (2)
4 first order ODE's.

Introduce 4 new state variables.

$$x_1, x_2, x_3, x_4.$$

we want the following

$$\dot{x}_1 = M_1(t, x_1, x_2, x_3, x_4)$$

$$\dot{x}_2 = M_2(t, x_1, x_2, x_3, x_4)$$

$$\dot{x}_3 = M_3(t, x_1, x_2, x_3, x_4)$$

$$\dot{x}_4 = M_4(t, x_1, x_2, x_3, x_4).$$

Given

$$x'' = 2y' + x - \frac{\tilde{m}(x+\mu)}{r_1^3} - \frac{\mu(x-\tilde{m})}{r_2^3} - fx'$$

$$y'' = -2x' + y - \frac{\tilde{m}y}{r_1^3} - \frac{\mu y}{r_2^3} - fy'$$

let

$x_1 = x$
$x_2 = x'$
$x_3 = y$
$x_4 = y'$

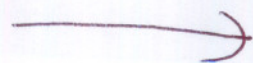
take
derivative \rightarrow

$$\dot{x}_1 = x_2$$

$$\begin{aligned} \dot{x}_2 &= x'' = 2y' + x - \frac{\tilde{m}(x+\mu)}{r_1^3} - \frac{\mu(x-\tilde{m})}{r_2^3} - fx' \\ &= 2x_4 + x_1 - \frac{\tilde{m}(x_1+\mu)}{r_1^3} - \frac{\mu(x_1-\tilde{m})}{r_2^3} - fx_2 \end{aligned}$$

$$\dot{x}_3 = x_4$$

$$\begin{aligned} \dot{x}_4 &= y'' = -2x' + y - \frac{\tilde{m}y}{r_1^3} - \frac{\mu y}{r_2^3} - fy' \\ &= -2x_2 + x_3 - \frac{\tilde{m}x_3}{r_1^3} - \frac{\mu x_3}{r_2^3} - fx_4 \end{aligned}$$



$$\text{where } r_1^3 = \left((x + \mu)^2 + y^2 \right)^{3/2} \quad (3)$$

$$\Downarrow$$

$$r_1^3 = \left((x_1 + \mu)^2 + x_3^2 \right)^{3/2}$$

$$\text{and } r_2^3 = \left((x - \tilde{\mu})^2 + y^2 \right)^{3/2}$$

$$\Downarrow$$

$$r_2^3 = \left((x_1 - \tilde{\mu})^2 + x_3^2 \right)^{3/2}$$

hence finally we write

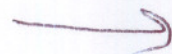
$$\dot{x}_1 = x_2 \quad (1)$$

$$\dot{x}_2 = 2x_4 + x_1 - \frac{\tilde{\mu}(x_1 + \mu)}{\left((x_1 + \mu)^2 + x_3^2 \right)^{3/2}} - \frac{\mu(x_1 - \tilde{\mu})}{\left((x_1 - \tilde{\mu})^2 + x_3^2 \right)^{3/2}} - f x_2 \quad (2)$$

$$\dot{x}_3 = x_4 \quad (3)$$

$$\dot{x}_4 = -2x_2 + x_3 - \frac{\tilde{\mu} x_3}{\left((x_1 + \mu)^2 + x_3^2 \right)^{3/2}} - \frac{\mu x_3}{\left((x_1 - \tilde{\mu})^2 + x_3^2 \right)^{3/2}} - f x_4 \quad (4)$$

these are the 4 1st order ODE (coupled) to solve.



Note that x_1 represents the x -coordinates of Apollo, x_3 represents the y -coordinates of Apollo.

So these are the coordinates to use to monitor position of satellite to see how close to earth it gets.

Solving using Runge-Kutta order 4. (RK4)

Example 1 This example shows how to use RK4 to solve one 1st order ODE.

$$\frac{dx}{dt} = f(t, x) \quad x = x_0 \text{ at } t = 0$$

$x = x_0; t = 0, \Delta t = 0.1, \text{NSTEPS} = 100$ (say).

→ DO FOR $I = 1, \text{NSTEPS}$

$$K_1 = f(t, x)$$

$$K_2 = f\left(t + \frac{\Delta t}{2}, x + \frac{\Delta t}{2} K_1\right)$$

$$K_3 = f\left(t + \frac{\Delta t}{2}, x + \frac{\Delta t}{2} K_2\right)$$

$$K_4 = f(t + \Delta t, x + \Delta t K_3)$$

$$x = x + \frac{\Delta t}{6} (K_1 + 2K_2 + 2K_3 + K_4)$$

$$t = t + \Delta t$$

WRITE(*,*) t, x

← END DO



Example 2 Use RK4 to solve 2 1st order (5)

Coupled ODE's.

$$\frac{dx}{dt} = f(t, x, y)$$

$$x = x_0 \quad t = 0$$

$$\frac{dy}{dt} = g(t, x, y)$$

$$y = y_0 \quad t = 0$$

Since these are coupled, we must determine the K's in locked-step fashion. as follows:

$$x = x_0$$

$$y = y_0$$

$$t = 0$$

$$NSTEPS = 100$$

$$\Delta t = 0.1$$

→ DO I=1, NSTEPS

$$K_{1,x} = f(t, x, y)$$

$$K_{1,y} = g(t, x, y)$$

$$K_{2,x} = f\left(t + \frac{\Delta t}{2}, x + \frac{\Delta t}{2} K_{1,x}, y + \frac{\Delta t}{2} K_{1,y}\right)$$

$$K_{2,y} = g\left(t + \frac{\Delta t}{2}, x + \frac{\Delta t}{2} K_{1,x}, y + \frac{\Delta t}{2} K_{1,y}\right)$$

$$K_{3,x} = f\left(t + \frac{\Delta t}{2}, x + \frac{\Delta t}{2} K_{2,x}, y + \frac{\Delta t}{2} K_{2,y}\right)$$

$$K_{3,y} = g\left(t + \frac{\Delta t}{2}, x + \frac{\Delta t}{2} K_{2,x}, y + \frac{\Delta t}{2} K_{2,y}\right)$$

$$K_{4,x} = f(t + \Delta t, x + \Delta t K_{3,x}, y + \Delta t K_{3,y})$$

$$K_{4,y} = g(t + \Delta t, x + \Delta t K_{3,x}, y + \Delta t K_{3,y})$$

$$t = t + \Delta t$$

$$x = x + \frac{\Delta t}{6} (K_{1,x} + 2K_{2,x} + 2K_{3,x} + K_{4,x})$$

$$y = y + \frac{\Delta t}{6} (K_{1,y} + 2K_{2,y} + 2K_{3,y} + K_{4,y})$$

So in our problem we do the same
except we have 4 1st order ODE'S.

⑥

$$\frac{dx_1}{dt} = M1(t, x_1, x_2, x_3, x_4)$$

$$\frac{dx_2}{dt} = M2(t, x_1, \dots)$$

$$\vdots$$

$$\frac{dx_4}{dt} = M4(\dots)$$

where $M1(t, x_1, x_2, x_3, x_4)$ is
RHS function $\rightarrow x_2$ in this
Case.

$M2(t, x_1, \dots)$ is from
before see eq (2)

$M3$ see eq (3)

$M4$ see eq (4)

I will show how to find K_1 and K_2 for this
problem.

~~$K_1 = M_1$~~ ~~$x_1=0, x_2=0, x_3=0$~~

$x_1 = 1.2; x_2 = 0, x_3 = 0, x_4 = -1.04935751$

$\Delta t = 0.1$

$NSTEPS = \frac{1.5 * T}{\Delta t}$ \checkmark Rounded. or $\frac{2 * T}{\Delta t}$ rounded

(i.e pick enough NSTEPS to see more
than one complete period.)



→ DO I = 1, NSTEPS

(7)

$$K_{1,x_1} = M1(t, x_1, x_2, x_3, x_4)$$

$$K_{1,x_2} = M2(-----)$$

$$K_{1,x_4} = M4(-----)$$

$$K_{2,x_1} = M1\left(t + \frac{\Delta t}{2}, x_1 + \frac{\Delta t}{2} K_{1,x_1}, x_2 + \frac{\Delta t}{2} K_{1,x_2}, x_3 + \frac{\Delta t}{2} K_{1,x_3}, x_4 + \frac{\Delta t}{2} K_{1,x_4}\right)$$

$$K_{2,x_2} = M2\left(t + \frac{\Delta t}{2}, x_1 + \frac{\Delta t}{2} K_{1,x_1}, x_2 + \frac{\Delta t}{2} K_{1,x_2}, \dots\right)$$

$$K_{2,x_4} = M4\left(t + \frac{\Delta t}{2}, x_1 + \frac{\Delta t}{2} K_{1,x_1}, x_2 + \frac{\Delta t}{2} K_{1,x_2}, \dots\right)$$

$$K_{3,x_1} = M1\left(t + \frac{\Delta t}{2}, x_1 + \frac{\Delta t}{2} K_{2,x_1}, x_2 + \frac{\Delta t}{2} K_{2,x_2}, \dots\right)$$

$$x_1 = x_1 + \frac{\Delta t}{6} (K_{1,x_1} + 2K_{2,x_1} + 2K_{3,x_1} + K_{4,x_1})$$

$$x_4 = x_4 + \frac{\Delta t}{6} (K_{1,x_4} + 2K_{2,x_4} + 2K_{3,x_4} + K_{4,x_4})$$

$$t = t + \Delta t$$

END DO

← here print

t, x₁, x₃

↓
x coordinate

↘
y coordinates

So you need to define 4 functions.

(1)

```
REAL FUNCTION M1(t, x1, x2, x3, x4)
  M1 = x2
  RETURN
END
```

$\mu = 1/82.45$ → $\bar{\mu} = 1 - \mu$; $f = 0$
REAL FUNCTION M2(t, x1, x2, x3, x4)
 $M2 = 2 * x4 + x1 - \bar{\mu} * (x1 + \mu) / ((x1 + \mu) * x2 + x3 * x2) * (3/2)$

— etc. see eq. (2)

```
RETURN
END
```

Same for M3, M4 Functions.

change $f = 0$ or $f = 1$ or $f = 0.1$

to do case (a), (b), (c).

Note about units

it is better to use the normalized units given.

i.e. $1 \equiv$ distance from earth center to moon center.

we are told this distance is 238,000 miles.

hence radius of earth is $\frac{4000}{238,000} = 0.0168067$

in Normalized units.

$$\frac{4000}{238,000}$$



To determine apollo distance from earth: ⑨

As you enter the loop, at the end of each loop iteration you'll get a new (x, y) coordinates for apollo. From then you can calculate r_1 which is distance of apollo to center of earth.

hence distance of apollo from surface of earth

is

$$r_1 - d$$

$$\text{where } d = 0.0168067$$